3. Form Tracking Number: Approving Civil Aviation Authority/Country **AUTHORIZED RELEASE CERTIFICATE** SE2R118019 FAA/UNITED STATES FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG 5. Work Order/Contract/Invoice Number: Organization Name and Address: AVMATS Component Support, 827 Texas Court, O'Fallon, Missouri 63366, SE2R158L 118019 6. Item: 7. Description: 9. Quantity: 10. Serial Number: 11. Status/Work: 8. Part Number: ESU 7375-0024 Repaired 12. Remarks: The work described in this block was performed for : PIUS PARTS On Reference Number: 147 The work specified has been accomplished in accordance with: Hamilton Sundstrand ESU repaired by performing alignment procedure. Unit repaired and tested in accordance with CMM 49-61-07, revision 1, dated 01/Dec/2001. No AD notes or SB's complied with at this time unless noted above. Full details on file with W.O. 118019 Other regulation specified in Block 12 14a. 14 CFR 43.9 Return to Service. 13a. Certifies the items identified above were manufactured in conformity to: Certifies that unless otherwise specified in Block 12, thework identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service. Approved design data and are in a condition for safe operation. Non-approved design data specified in Block 12. 14c. Approval/Certificate No: 13b. Authorized Signature: 13c. Approval/Authorization No: SE2R158L 13d. Name (Typed or Printed): 13e. Date (dd/mmm/yyyy): 14e. Date (dd/mmm/yyyy): 25/Oct/2023 It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

NSN: 0052-00-012-9005

^{*} Installer must cross-check eligibility with applicable technical data.

WORK ORDER

AVMATS Component Support 827 Texas Court O'Fallon, Missouri

FAA Approved Repair Station #SE2R158L EASA Reference Number: EASA 145.4353

NON EASA COMPLIANT

Address:

Work Order: 118019

Customer: PIUS PARTS

MAIN 11 BEE TREE CIRCLE

ROCKPORT, TEXAS 78382 United States

1.00 Part #: 161043-401

Serial #: 7375-0024

Qty: ESU Item: 1

Squawk: 1.00

Manufacturer:

Resolution:

Discrepancy: Function test and evaluate for repair if needed.

Hamilton Sundstrand ESU repaired by performing alignment procedure. Unit repaired and tested in accordance with CMM 49-61-07, revision 1, dated 01/Dec/2001, --END--

Department: Electrical

P.O. #: 147

Created: 9/12/2023

Job #:

Date Completed Technician > Water Ruh 25 OCT 2023 Michael The air craft and / or component identified above was repaired and inspected in accordance with current Federal Aviation Agency Regulations and was found airworthy and is approved for return to service. Date 25 Oct 2023 FAA Certified Repair Station #SE2R158L EASA Reference Number: EASA 145.4353 for AVMATS Component Support 827 Texas Court, O'Fallon Missouri 63366

avCPR_WorkOrderDetail.rpt

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Printed: 10/25/2023 8:39:45AM

| BITE # | DECODED BITE INFORMATION | 164279-1 |
|--------|---|-----------|
| | RESET START FUEL VALVE & EXCITER SIGNAL OUT (5%) MAIN FUEL VALVE SIGNAL OUT (14%) START FUEL VALVE & EXCITER SIGNAL OFF (70%) 90% RPM SWITCH ON | START |
| | READY FOR SERVICE (90% + 10 SEC) PROCESSOR BOARD FAILURE DATA SYSTEM FAILURE OVERTEMPERATURE PROBE # 2 OVERSPEED | OPERATION |
| | UNDERSPEED FAIL TO START OIL SYSTEM FAILURE SHORTED OIL PRESSURE SW. | |
| | BLEED AIR VALVE OPEN SHORTED THERMOCOUPLE PROBE # 2 OVERTEMP PROBE # 1/OPEN THERMOCOUPLE # 1 OPEN THERMOCOUPLE PROBE # 2 | |
| | PROCESSOR SEQUENCE FAIL NO TEMP DATA | |