

CONFIGURATION MODULE
P/N 17680-01
REF. ENG. DWG. 400-017680

WT: 1 LB
TSO-C115a

S/N **316**

MOD
1 2 3 4

AlliedSignal Avionics Inc.
PRESCOTT, ARIZONA 86301 U.S.A.

WARN
WARRANT
WHEN TH
IS BRO

300-317603-71 REV. B
23720
GMDD

GNS-X/ES
P/N 17450-0101
REF. ENG. DWG. 400-017450

OPTIONS	S/N	WT	LBS
0002	212	9.8	

THIS LRU CAN BE REPLACED WITH AN LRU OF THE SAME MOD STATUS, OR HIGHER NUMBER SOFTWARE MOD STATUS.

THIS LRU CAN BE REPLACED WITH AN LRU HAVING THE SAME SET OF HARDWARE MODS. ADDITIONAL MODS. ARE ACCEPTABLE.

SOFTWARE MODS															
1	2	3	4	5	6	7	8								
9	10	11	12	13	14	15	16								
17	18	19	20	21	22	23	24								
25	26	27	28	29	30	31	32								

HARDWARE MODS							
1	2	3	4	5	6	7	8
9	10	11	12	13	14	15	16

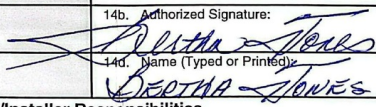
Software Level = 2

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TSO-C115a

DO-160C ENVIRONMENTAL CATEGORIES
[D1F1] VBA-(MNB) E1 XXXXX ZAAZZRZKXX

SE 00 00
FT
CRL

1. Approving Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG		3. Form Tracking Number: 050115-20140710-777	
4. Organization Name and Address: Honeywell International, Inc. One Technology Center 23500 West 105th Street Olathe, Kansas 66061		Repair Station PR2R093L		5. Work Order/Contract/Invoice Number: AP0422	
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status / Work:
001	GNS-XES	17450-0101-0002	1	144	REPAIRED
12. Remarks: THE SERVICE SPECIFIED HAS BEEN ACCOMPLISHED IN ACCORDANCE WITH: Repaired per OEM document/Drawings GNS-XES 149-117988 Rev 8 10-01-1997. Repaired per OEM documents/Drawings 300-317637 Rev C. 310-017627-01 Rev E 11-01-1995 COM 17450-0101-0002 Rev 9, 09-dec-1996 ATE 222-00411-0003 Rev -, 6/9/1998 SEE ATTACHED DOCUMENTS AS APPLICABLE FOR WORK PERFORMED HONEYWELL CERTIFIES THAT THE WORK SPECIFIED IN BLOCKS 11/12 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART 145 AND IN RESPECT TO THAT WORK THE [PRODUCT/ARTICLE] IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA PART 145 APPROVAL NO. EASA.145.4140					
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation <input type="checkbox"/> Non-approved design data specified in Block 12			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.		
Authorized Signature:		13c. Approval/Authorization No.:	14b. Authorized Signature:		14c. Approval/Certificate No.:
					PR2R093L
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):	14d. Name (Typed or Printed):		14e. Date (dd/mm/yyyy):
			DEBRA JONES		18/Jul/2014
User/Installer Responsibilities					
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs works in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.					