

**CONFIGURATION MODULE**  
 P/N 17680-01  
 REF. ENG. DWG. 400-017680

WT: 1 LB  
 TSO-C115a

S/N  
 316

MOD  
 1 2 3 4

AlliedSignal Avionics Inc.  
 PRESCOTT, ARIZONA U.S.A.

WARN  
 WARRANT  
 WHEN TH  
 IS BRO

300-317603-71 REV. B  
 23720  
 GMDD

**GNS-X/ES**  
 P/N 17450-0101  
 REF. ENG. DWG. 400-017450

OPTIONS	S/N	WT	LBS
0002	212	9.8	

HARDWARE MODS  
 1 2 3 4 5 6 7 8  
 9 10 11 12 13 14 15 16

THIS LRU CAN BE REPLACED WITH AN LRU HAVING THE SAME SET OF HARDWARE MODS. ADDITIONAL MODS ARE ACCEPTABLE.

SOFTWARE MODS  
 1 2 3 4 5 6 7 8  
 9 10 11 12 13 14 15 16  
 17 18 19 20 21 22 23 24  
 25 26 27 28 29 30 31 32

THIS LRU CAN BE REPLACED WITH AN LRU OF THE SAME MOD STATUS, OR HIGHER NUMBER SOFTWARE MOD STATUS.



Software Level = 2

AlliedSignal Avionics Inc.  
 PRESCOTT, ARIZONA 86301 U.S.A.

TSO-C115a

DO-160C ENVIRONMENTAL CATEGORIES  
 [D1F1] VBA-(MNB) E1 XXXXX ZAAZZRZKXX

SEP 00 2003  
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 CRL

1. Approving Civil Aviation Authority/Country: <b>FAA/United States</b>		2. <b>AUTHORIZED RELEASE CERTIFICATE</b> FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 050115-20140710-777	
4. Organization Name and Address: Honeywell International, Inc. One Technology Center 23500 West 105th Street Olathe, Kansas 66061		Repair Station PR2R093L		5. Work Order/Contract/Invoice Number: AP0422		
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status / Work:	
001	GNS-XES	17450-0101-0002	1	144	REPAIRED	
12. Remarks: THE SERVICE SPECIFIED HAS BEEN ACCOMPLISHED IN ACCORDANCE WITH: Repaired per OEM document/Drawings GNS-XES 149-117988 Rev 8 10-01-1997. Repaired per OEM documents/Drawings 300-317637 Rev C. 310-017627-01 Rev E 11-01-1995 COM 17450-0101-0002 Rev 9, 09-dec-1996 ATE 222-00411-0003 Rev -, 6/9/1998  SEE ATTACHED DOCUMENTS AS APPLICABLE FOR WORK PERFORMED HONEYWELL CERTIFIES THAT THE WORK SPECIFIED IN BLOCKS 11/12 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART 145 AND IN RESPECT TO THAT WORK THE [PRODUCT/ARTICLE] IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA PART 145 APPROVAL NO. EASA.145.4140						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation <input type="checkbox"/> Non-approved design data specified in Block 12			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature: 		13c. Approval/Authorization No.: PR2R093L		14b. Authorized Signature: 		14c. Approval/Certificate No.: PR2R093L
13d. Name (Typed or Printed): D. JONES		13e. Date (dd/mmm/yyyy): 18 Jul 2014		14d. Name (Typed or Printed): D. JONES		14e. Date (dd/mmm/yyyy): 18 Jul 2014
<b>User/Installer Responsibilities</b>						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs works in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						