



A1 A2

RECOMMENDED INSTALLATION TORQUE 90 IN-LBS MAX.

SM400D68

CAUTION SHORT X1 TO X2 BEFORE HIGH VOLTAGE TEST

EATON

RECOMMENDED INSTALLATION TORQUE 90 IN-LBS MAX.



A1 MFD9934 A2

RECOMMENDED INSTALLATION
TORQUE 90 In-lbs MAX.

SM400D68

CAUTION SHORT +X1 TO -X2 BEFORE
HIGH VOLTAGE TEST

EAT•N

RECOMMENDED INSTALLATION
TORQUE 19 In-lbs MAX

+X1

IND

-X2

1. APPROVING CIVIL AVIATION AUTHORITY/COUNTRY: **FAA / UNITED STATES**

2. **AUTHORIZED RELEASE CERTIFICATE**
 FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

3. Form Tracking Number: **254957**

4. Organization Name and Address: **CONSOLIDATED AIRCRAFT SUPPLY CO. INC. (631) 981-7700 800-422-6300
 55 RAYNOR AVENUE, RONKONKOMA, NEW YORK 11779 FAX: (631) 981-7706
 FAA#G11R167K**

5. Work Order/Contract/Invoice Number: **P00582**

6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:
1	RELAY	SM400D68	1	00013	INSPECTED

12. Remarks: 1. THIS UNIT ~~WAS~~ FUNCTIONALLY TESTED IN ACCORDANCE WITH MANUFACTURER'S CURRENT MANUAL.
 2. THE WORK PERFORMED IS DETAILED ON THE ATTACHED INSPECTION REPORT.
 3. CERTIFIES THAT THE WORK SPECIFIED IN BLOCKS 11 AND 12 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART-145 AND IN RESPECT TO THAT WORK THE COMPONENT IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA PART-145 APPROVAL NUMBER: EASA 145 4346.

13a. Certifies the items identified above were manufactured in conformity to:
 Approved design data and are in condition for safe operation
 Non-approved design data specified in Block 12.

14a. 14 CFR 43.9 Return To Service Other Regulation Specified in Block 12
 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, Part 43 and in respect to that work, the items are approved for return to service.

13b. Authorized Signature: _____

13c. Approval/Authorization No: _____

14b. Authorized Signature: **[Signature]**

14c. Approval/Certificate No: **G11R167K**

13d. Name (typed or printed): _____

13e. Date: (dd/mm/yyyy) _____

14d. Name (typed or printed): **CAR MATW**

14e. Date (dd/mm/yyyy): **03/SEP/2025**

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.

Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.