



UNISON
IGNITION EXCITER

DANGER
- HIGH OUTPUT VOLTAGE AND AMPERAGE
- DISCONNECT INPUT CURRENT AND WAIT ONE MINUTE BEFORE REMOVING OUTPUT LEADS
- OPERATE ONLY WITH LEADS AND IGNITER PLUGS CONNECTED.

INPUT
PIN 1 CH A +28V
3 CH B +28V
6 BIT A
7 DAU A
8-9 GND
10 BIT B
11 DAU B
13 DEEC

CUST NO: 99193-SOCN-3061186-1
 MFR59501 UNISON TYPE: TX310
 PART NO: 9060280-1 C
 AMPS: 5.0 MAX
 DUTY: CONTINUOUS
 S/N: 99327985
 VOLTS: 10-30 VDC
 FAA-PMA
 U.S. PATENTS: 5155437, 5343154, 5347422



1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 26952625-18995736-1-1	
4. Organization Name and Address: Unison Industries L.L.C. 7375 Baymeadows Way Jacksonville, FL 32256-7525		REPAIR STATION FACILITY CERTIFICATION NO. UILR299K			5. Work Order/Contract/Invoice Number: REPAIR NUMBER 9497541 P.O. 72730	
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:	
1	EXCITER,IGN TYPE TX310	9060280-1	1	99327985	INSPECTED/TESTED	
12. Remarks: TESTED AND INSPECTED PER DWG 9060280-1 REV C DTD NOV/08/1996 AND SPEC JS0771 REV E DTD APR/30/1999 For European shipments: Certifies that work specified in block 11/12 was carried out in accordance with EASA Part-145 and in respect to that work the aircraft component is considered ready for release to service under EASA Part-145 Approval Number: EASA 145.4976. For Canadian Shipments: Certifies the work described in blocks 11/12 is released under the terms of the US - Canada Bilateral Aviation Safety Agreement. Details on file under W.O. number listed in block 5. No AD/ SB incorporated unless stated in block 12 remarks						
13a. Certify the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.				14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part-43 and in respect to that work, the items are approved for return to service.		
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature: <i>Lpe</i>		
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14c. Approval/Certificate No.: UILR299K		
				14d. Name (Typed or Printed): Rosa, LuisOmar		
				14e. Date (dd/mm/yyyy): 15/FEB/2026		
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						