



KCU 951 MODS 1 2 3 4 5 6 7 8
HF CONTROLLER

FCC ID: ASY90QKHF950
KING RADIO CORP.

P/N 064-1016-03 S/N 3 6 4 2
TSO C31c, C32c 00 163
ENV.CAT.00 160 DIA/KPS/XXXXXXABAAB
KING RADIO CORP OLATHE, KANS. U.S.A. 66061 WT. 1.1 LBS.(0.50 KG)

1. Approving National Aviation Authority/Country FAA/UNITED STATES	2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	3. Form Tracking Number: 006YA0001001001
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4. Organization Name and Address: Duncan Aviation/Lincoln Airport/Lincoln, NE 68524 JGVR194F	5. Work Order/Contract/Invoice Number: 006YA
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6. Item:	7. Description:	8. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:
1	HF CONTROL	064-1016-03	N/A	1	3642	REPAIRED

13. Remarks:

Customer: AIRMARK INTERNATIONAL

Discrepancy: Squelch knob missing.

Preliminary Findings: Bench tested and found operational, but missing squelch knob.

Corrective Actions: Replaced squelch knob. Function tested unit per HONEYWELL KCU 951 Control Unit Maintenance Manual, Revision 1, dated 04/01/2006.

Duncan Aviation certifies that the work specified in Blocks 12 and 13 was performed I.A.W. EASA Implementation Rule part 145 approval, and with respect to that work, the aircraft component is considered ready for release to service under EASA approval number 145.4392.

All EASA/FAA Airworthiness Directives are the responsibility of the installer.

No current U.S. Airworthiness Directives apply.

Technician: JAMES ARTHUR BROCKMAN

This document constitutes a signed copy of the work order

Approval for return to service

See attached parts list

14. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in a condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 13.	19. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input checked="" type="checkbox"/> Other regulations specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.
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15. Authorized Signature:	16. Approval Authorization No.:	20. Authorized Signature: <i>[Signature]</i> DUNCAN QI 286	21. Approval/Certificate No.: JGVR194F
17. Name (Type or Printed):	18. Date (m/d/y):	22. Name (Type or Printed): RODNEY L. WALTHER	23. Date (m/d/y): Jun 17, 2010

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.

Statements in blocks 14 and 19 do not constitute installations certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.