



A1

A2

RECOMMENDED INSTALLATION
TORQUE 90 In-lbs MAX.

SM400D68

CAUTION SHORT +X1 TO -X2 BEFORE
HIGH VOLTAGE TEST

EAT•N

RECOMMENDED INSTALLATION
TORQUE 19 In-lbs MAX.

+X1

IND

-X2



PIUS PARTS

1. APPROVING CIVIL AVIATION AUTHORITY/COUNTRY: FAA / UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 261436	
4. Organization Name and Address: CONSOLIDATED AIRCRAFT SUPPLY CO. INC. (631) 981-7700 800-422-6300 55 RAYNOR AVENUE, RONKONKOMA, NEW YORK 11779 FAX: (631) 981-7706				5. Work Order/Contract/Invoice Number: P00961		
FAA#G11R167K						
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:	
1	RELAY	SM400D68	1	N/A	INSPECTED	
12. Remarks: 1. THIS UNIT WAS FUNCTIONALLY TESTED IN ACCORDANCE WITH MANUFACTURER'S CURRENT MANUAL. CASCO#4348 2. THE WORK PERFORMED IS DETAILED ON THE ATTACHED INSPECTION REPORT. 3. CERTIFIES THAT THE WORK SPECIFIED IN BLOCKS 11 AND 12 WAS CARRIED OUT IN ACCORDANCE WITH EASA PART-145 AND IN RESPECT TO THAT WORK THE COMPONENT IS CONSIDERED READY FOR RELEASE TO SERVICE UNDER EASA PART-145 APPROVAL NUMBER: EASA 145.4348 HW DWG SM400D68 REV E 2/24/99						
13a. Certifies the items identified above were manufactured in conformity to: <input type="checkbox"/> Approved design data and are in condition for safe operation. <input type="checkbox"/> Non-approved design data specified in Block 12.			14a. <input checked="" type="checkbox"/> 14 CFR 43.9 Return to Service <input type="checkbox"/> Other Regulation Specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, Part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:	13c. Approval/Authorization No.:	14b. Authorized Signature:	14c. Approval/Certificate No.: G11R167K			
13d. Name (typed or printed):	13e. Date (dd/mm/yyyy):	14d. Name (typed or printed): CART D'ATW	14e. Date (dd/mm/yyyy): 20/MAR/2026			
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						
FAA Form 8130-3 (02-14)				NSN: 0052-00-012-9005		